OPTICS REQUIREMENTS FOR THE GENERATION-X X-RAY TELESCOPE

S. L. O'Dell, R. F. Elsner, J. J. Kolodziejczak, B. D. Ramsey, M. C. Weisskopf NASA Marshall Space Flight Center (MSFC), 320 Sparkman Drive, Huntsville AL 35805, USA W. W. Zhang, D. A. Content, R. Petre, T. T. Saha

NASA Goddard Space Flight Center (GSFC), Greenbelt MD 20771, USA

P. B. Reid, D. A. Schwartz, R. J. Brissenden, M. Elvis, M. Freeman, T. Gaetz, P. Gorenstein, D. Jerius, M. Juda, S. S. Murray, W. A. Podgorski, S. J. Wolk *Harvard–Smithsonian Center for Astrophysics, 60 Garden Street, Cambridge MA 02138, USA* S. Trolier-McKinstry

Pennsylvania State University, 151 Material Research Lab., University Park PA 16802, USA.

US, European, and Japanese space agencies each now operate successful X-ray missions—NASA's *Chandra*, ESA's XMM-*Newton*, and JAXA's *Suzaku* observatories. Recently these agencies began a collaboration to develop the next major X-ray astrophysics facility—the *International X-ray Observatory* (IXO)—for launch around 2020. IXO will provide an order-of-magnitude increase in effective area (Table 1), while maintaining good (but not sub-arcsecond) angular resolution.

Mission Status Launch Aperture area Resolution Chandra (nee AXAF) Operating 1999 0.08 m^2 0.5" XMM-Newton Operating 1999 0.43 m^2 15" 120" Suzaku (nee Astro-E2) Operating 2005 0.18 m^2 International X-ray Observatory (IXO) ≈ 2020 Planning 3.5 m^2 ≤ 5" Generation-X Concept ≈ 2035 $50 \,\mathrm{m}^2$ ≈ 0.1″

Table 1. Comparison of X-ray telescopes

X-ray astronomy beyond IXO will require optics with even larger aperture areas and much better angular resolution. We are currently conducting a NASA strategic mission concept study to identify technology issues and to formulate a technology roadmap for a mission—*Generation-X* (*Gen-X*)—to provide these capabilities.

Achieving large X-ray collecting areas in a space observatory requires extremely lightweight mirrors. For (2-reflection) X-ray optics with graze angles of order 0.01 radian, the mirror surface area is about 200 times the aperture area. Thus, the Gen-X requirement for 50 m² aperture area implies 10000 m² of mirror surface area—i.e., 10

tonne of mirrors at an areal density of 1 kg m⁻². NASA's plan for the Ares V heavy-lift capability will enable the insertion of *Generation-X* into an Earth-Sun L2 (second Lagrange-point) orbit, in a single launch of a single observatory (Figure 1).

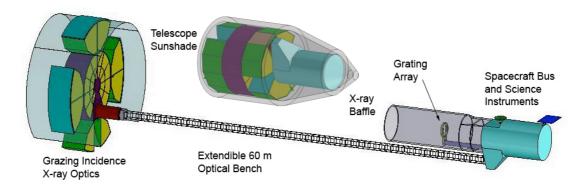


Figure 1. Conceptual configuration of the *Generation-X* telescope. The diagram shows the telescope stowed within an Ares-V 10-m-diameter shroud and deployed for in-space operation.

Achieving 0.1" X-ray imaging with lightweight mirrors presents a major technological challenge. Accomplishing this will require excellent mirror surfaces (≤0.1 µradian RMS deviations), precise alignment, and exceptional figure control to compensate for mounting stresses. Very likely, achieving and maintaining alignment and figure control will involve active X-ray optics (Figure 2).

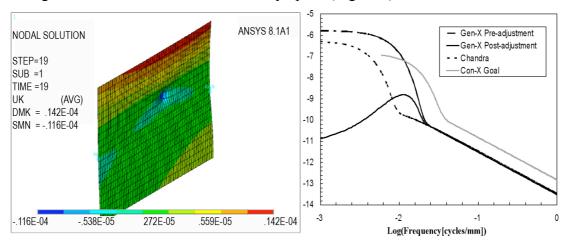


Figure 2. (Left) A finite-element analysis of the influence function for a piezoelectric bimorph zone on a thin mirror; (Right) schematic illustration of the suppression of low-frequency figure errors to correct a mounted mirror to meet the Gen-X requirements on imaging quality.

Contact address: <u>Steve.O'Dell@nasa.gov</u> (or <u>Stephen.L.ODell@nasa.gov</u>) NASA/MSFC/VP62, 320 Sparkman Drive, Huntsville AL 35805-1912, USA.